

AUV Modeling, Control, Obstacle avoidance

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Outlines

- ❑ Coordinate system, general 6Dof inertial and relative
- ❑ Modeling MIMO system using Fourier Integral method
- ❑ MIMO Control design
 - Specifications
 - Loop shaping
- ❑ Optimal path planner using the DIDO solver
- ❑ Experiments

Coordinate System

Technical terms (terminology)

x

* Surge: u , X

* Roll: p , L

y

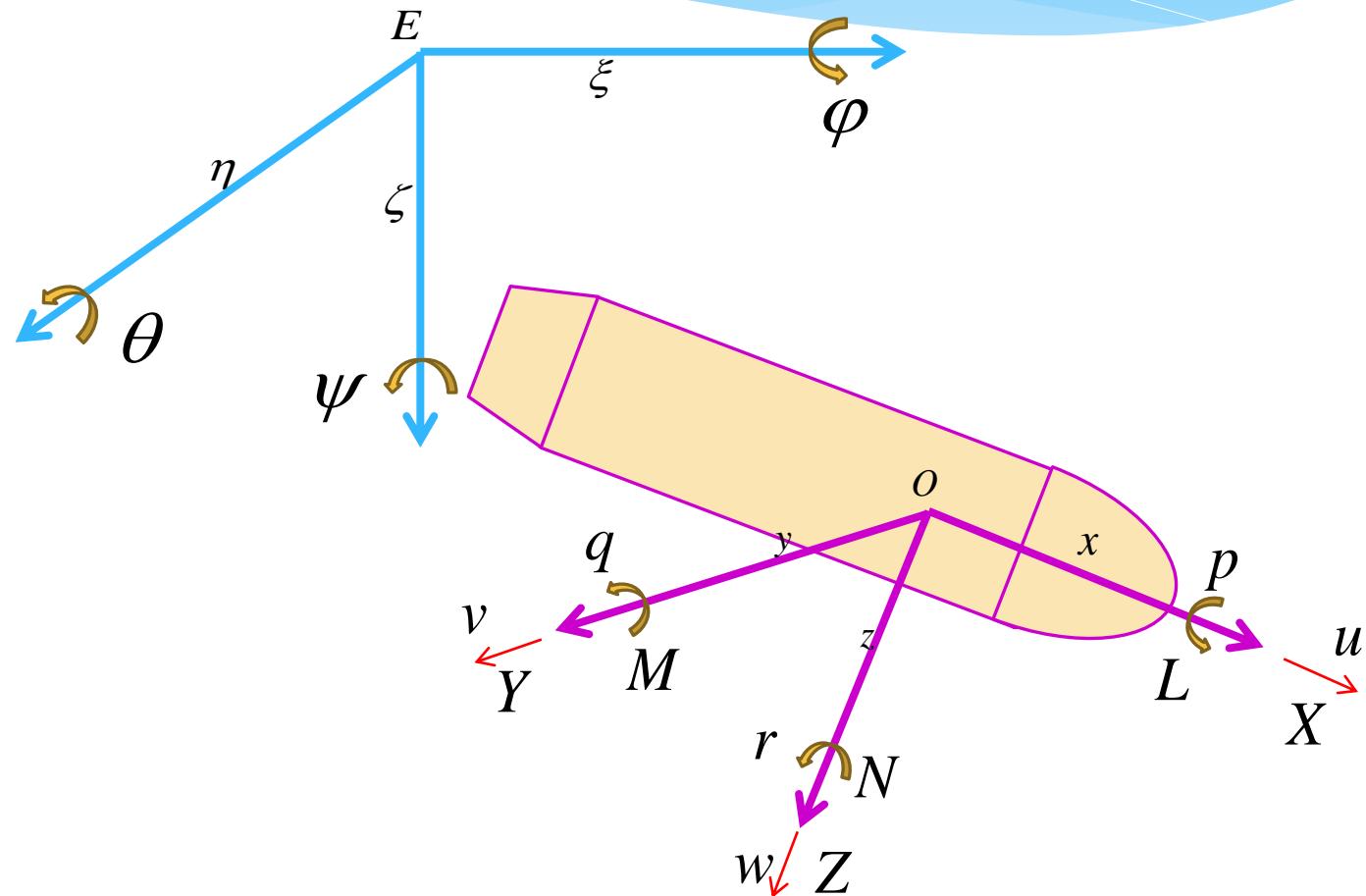
* Sway: v , Y

* Pitch: q , M

z

* Heave: w , Z

* Yaw: r , N



State Control

AUV State

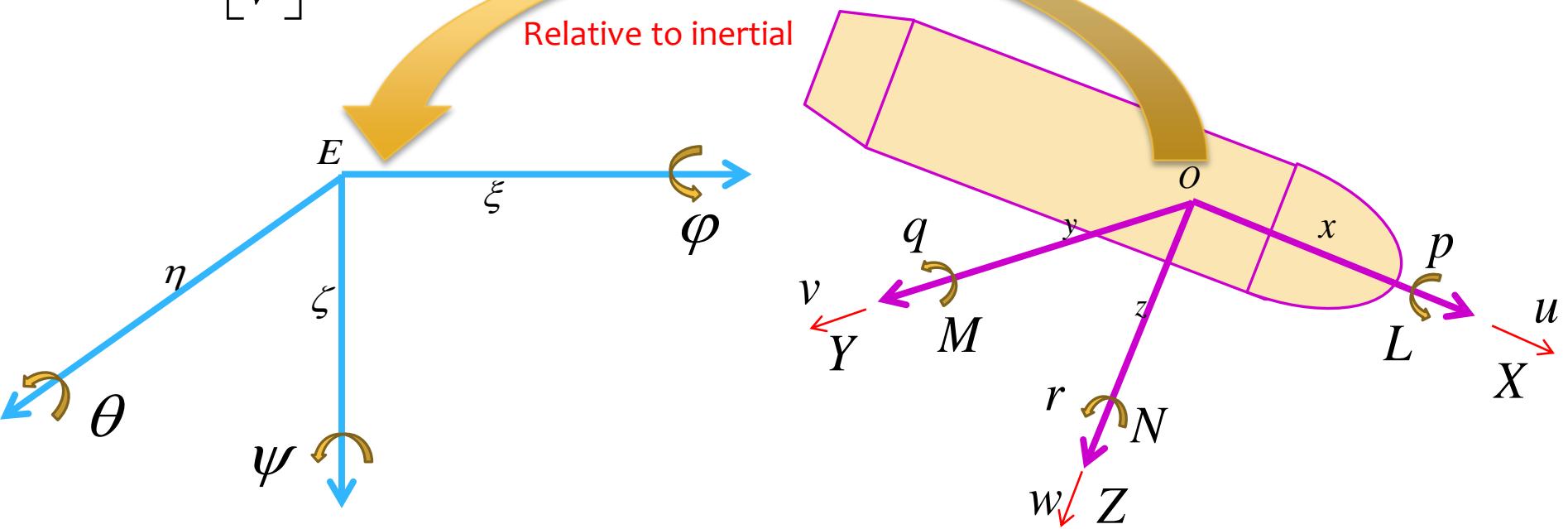
$$R = [r' \mid \Lambda']' = [\xi \ \eta \ \zeta \mid \varphi \ \theta \ \psi]'$$

$$V = [U' \mid \Omega']' = [u \ v \ w \mid p \ q \ r]'$$

State $\square \begin{bmatrix} R \\ V \end{bmatrix}$

Transformation

Relative to inertial



Transformation

- Transformation from fixed-body frame to inertial coordinate system

Inertial velocity

$$\begin{bmatrix} \dot{\xi}_G \\ \dot{\eta}_G \\ \dot{\zeta}_G \\ \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix} = \begin{bmatrix} T_1 & 0_3 \\ 0_3 & T_2 \end{bmatrix} \begin{bmatrix} u \\ v \\ w \\ p \\ q \\ r \end{bmatrix}$$

Relative velocity

- Transform matrixes

$$T_1 = \begin{bmatrix} \cos\psi \cos\theta & \cos\psi \sin\theta \sin\varphi - \sin\psi \cos\varphi & \cos\psi \sin\theta \cos\varphi + \sin\psi \sin\varphi \\ \sin\psi \cos\theta & \sin\psi \sin\theta \sin\varphi + \cos\psi \cos\varphi & \sin\psi \sin\theta \cos\varphi - \cos\psi \sin\varphi \\ -\sin\theta & \cos\theta \sin\varphi & \cos\theta \cos\varphi \end{bmatrix}$$

$$T_2 = \begin{bmatrix} 1 & \tan\theta \sin\varphi & \tan\theta \cos\varphi \\ 0 & \cos\varphi & -\sin\varphi \\ 0 & \sin\varphi \sec\theta & \cos\varphi \sec\theta \end{bmatrix}$$

Momentum Theorem

momentum \triangleq Mass \cdot Velocity (scalar product)

external forces and moments = $\frac{d}{dt}$ [momentum]

Origin is located at the center of buoyancy and the center of gravity lies at the point \mathbf{r}_G .

➤ Newton's equation of motion $\sum \mathbf{F} = m \mathbf{a}_G$

\mathbf{a}_G \triangleq center mass acceleration, $m \triangleq$ body mass , $\sum \mathbf{F} \triangleq$ external forces

$$\mathbf{a}_G = \frac{\partial \mathbf{U}}{\partial t} + \boldsymbol{\Omega} \times \mathbf{U} + \dot{\boldsymbol{\Omega}} \times \mathbf{r}_G + \boldsymbol{\Omega} \times \boldsymbol{\Omega} \times \mathbf{r}_G$$

$$\boldsymbol{\Omega} = [\mathbf{p}, \mathbf{q}, \mathbf{r}]' \quad , \quad \mathbf{U} = [\mathbf{u}, \mathbf{v}, \mathbf{w}]'$$

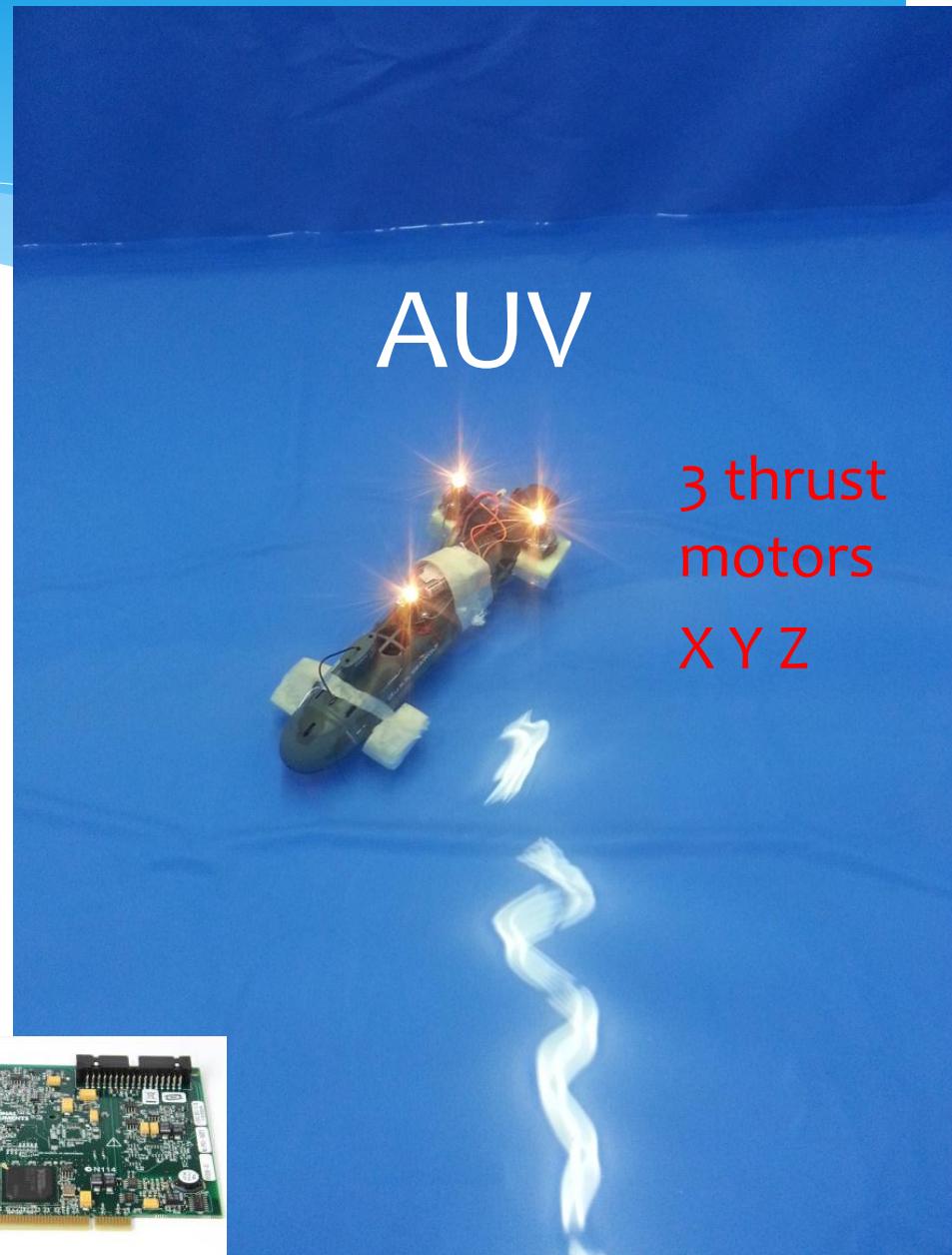
➤ Euler's equation of motion $\sum \mathbf{M}_B = \dot{\mathbf{H}}_G + \mathbf{r}_G \times m \mathbf{a}_G$

$\dot{\mathbf{H}}_G$ \triangleq rate of change of angular momentum about the center of gravity , $\sum \mathbf{M}_B \triangleq$ external moments

$$\dot{\mathbf{H}}_G = [\mathbf{I}] \dot{\boldsymbol{\Omega}} + \boldsymbol{\Omega} \times [\mathbf{I}] \boldsymbol{\Omega}$$

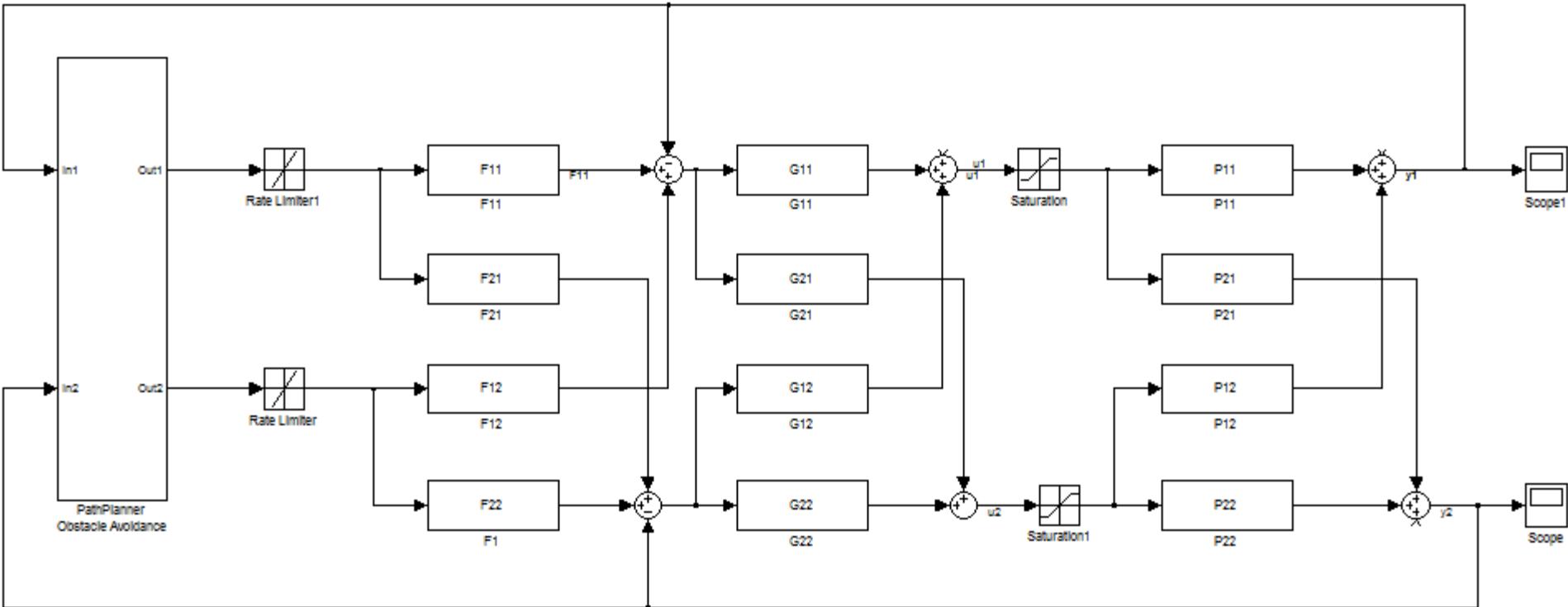
$[\mathbf{I}] \triangleq$ diagonal inertia matrix (diag $[I_x \ I_y \ I_z]$)

Laboratory Setup



Control Architecture

- ❑ 2 Dof MIMO:
- ❑ Path planner (MPC)
- ❑ F is the prefilter, G is the controller, P is the plant.
- ❑ Additions:
 - rate limit on the position reference command
 - Saturation on the control signal



Modeling

□ Modeling using Fourier integral method

□ Execute on all variations of u and y:

- From u_1 to y_1 and y_2
- From u_2 to y_2 and y_1

$$\square \text{Plant}(j\omega) = \delta$$

$$|\text{Plant}(j\omega)| = \frac{B}{A}$$

$$y_s = \frac{1}{nT} \int_0^{nT} y(t) \sin(\omega t) dt = \frac{1}{nT} \int_0^{nT} (B \sin(\omega t + \delta) + e(t)) \sin(\omega t) dt =$$

$$= \frac{1}{nT} \int_0^{nT} B \sin(\omega t + \delta) \sin(\omega t) dt + \frac{1}{nT} \int_0^{nT} e(t) \sin(\omega t) dt =$$

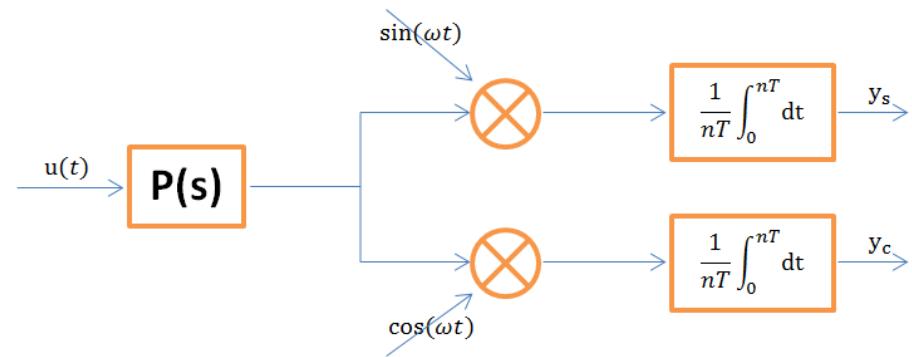
$$= \frac{B}{2} \cos(\delta) - \underbrace{\frac{B}{2nT} \int_0^{nT} \cos(\omega t + \delta) dt}_{=0} + \underbrace{\frac{1}{nT} \int_0^{nT} e(t) \sin(\omega t) dt}_{=0, \text{uncorrelated signals}}$$

$$y_s = \frac{B}{2} \cos(\delta)$$

$$y_c = \frac{B}{2} \sin(\delta)$$

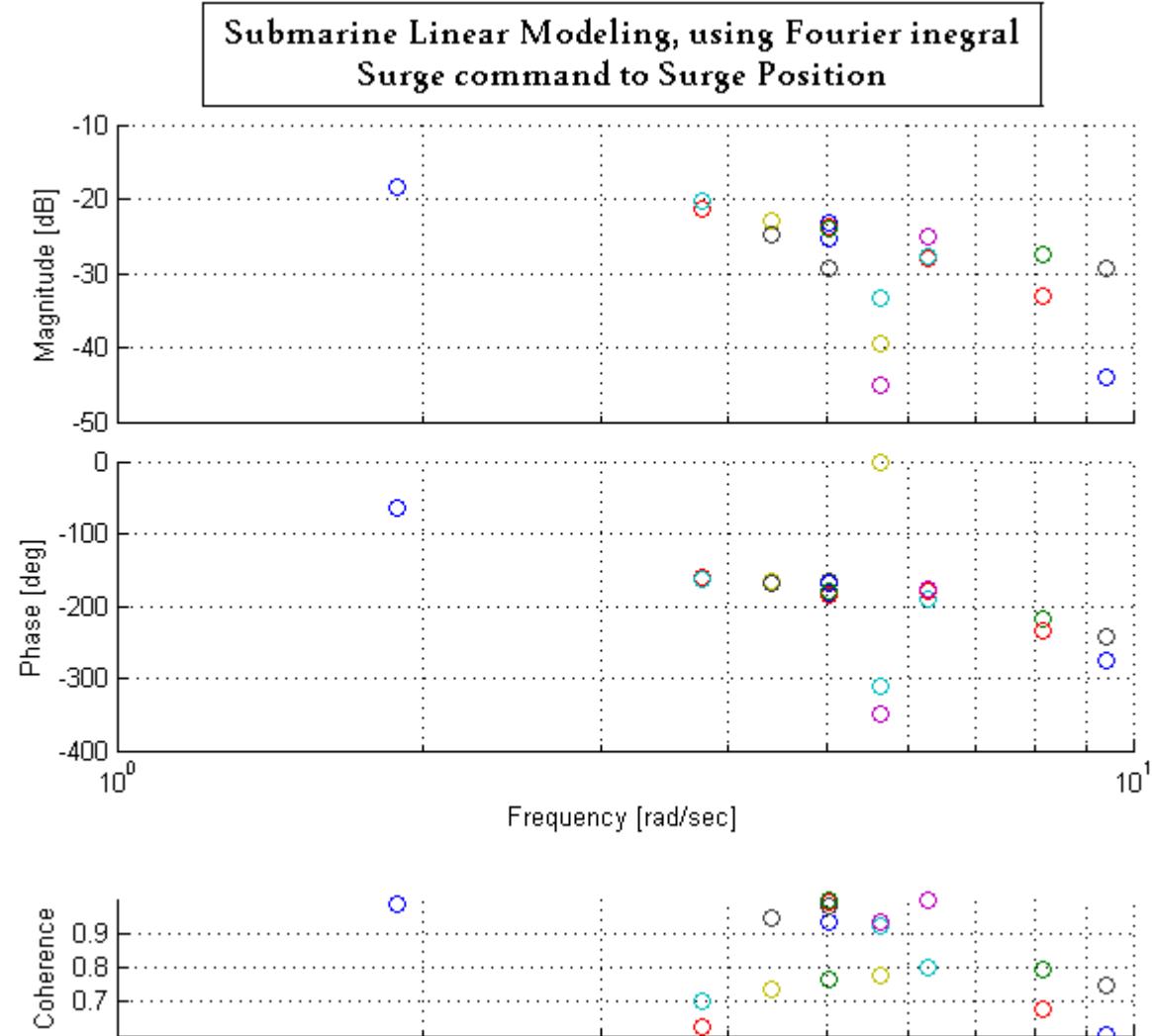
$$\delta = \arctan 2(y_s, y_c)$$

$$B = 2\sqrt{y_s^2 + y_c^2}$$



Modeling

- ❑ Modeling results
(1/4 of the plants)
- ❑ From Thrust motor
input to Longitude
position



Control Design

❑ Specifications:

- stabilize the close loop
- Step response
 - Settling time 10 [s]
 - Overshoot < 10%
- Disturbance rejection
 - 6 [dB] for all frequencies
- Cross coupling sensitivity
 - 3 [dB] for all frequencies

All time domain specifications translated into frequency domain using 2nd order transfer function formulations.

Control Design specifications

- Time domain specifications translated into frequency domain specifications using 2nd order formulations

$$G(s) = \frac{\omega_n}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$$T_{settling} = \frac{4}{\zeta\omega_n}, \text{ for } 2\% \text{ error}$$

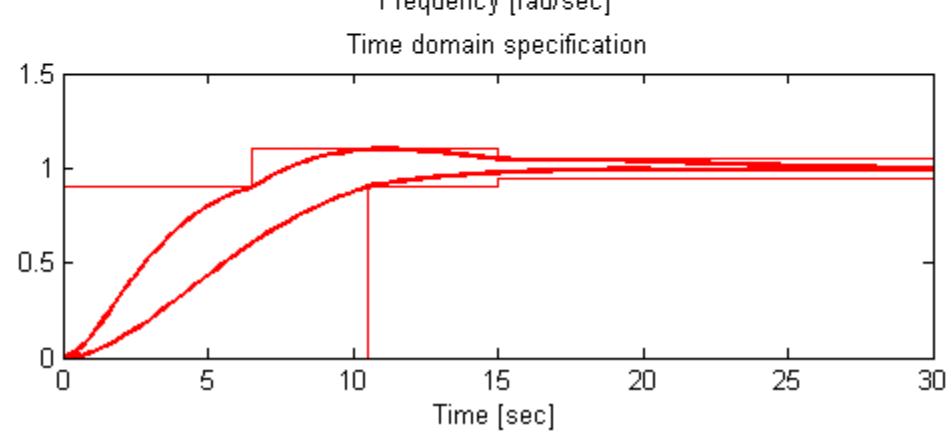
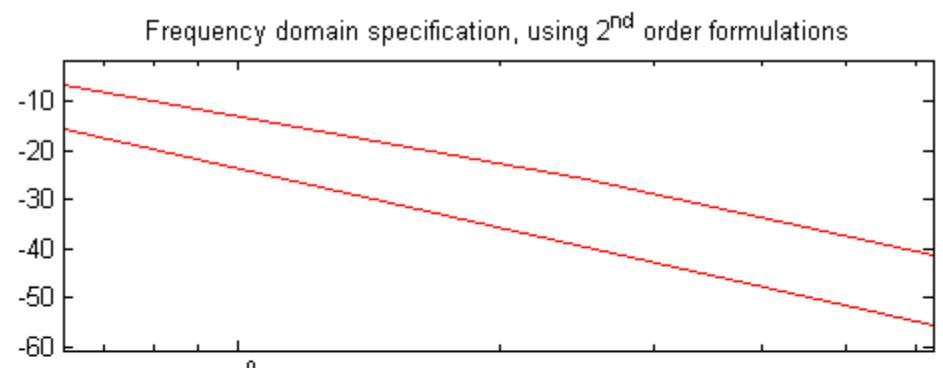
$$OS = 1 + e^{-\frac{\pi\zeta}{\sqrt{1-\zeta^2}}}$$

- Servo specification

$$a(\omega) \leq |ClosedLoop| \leq b(\omega)$$

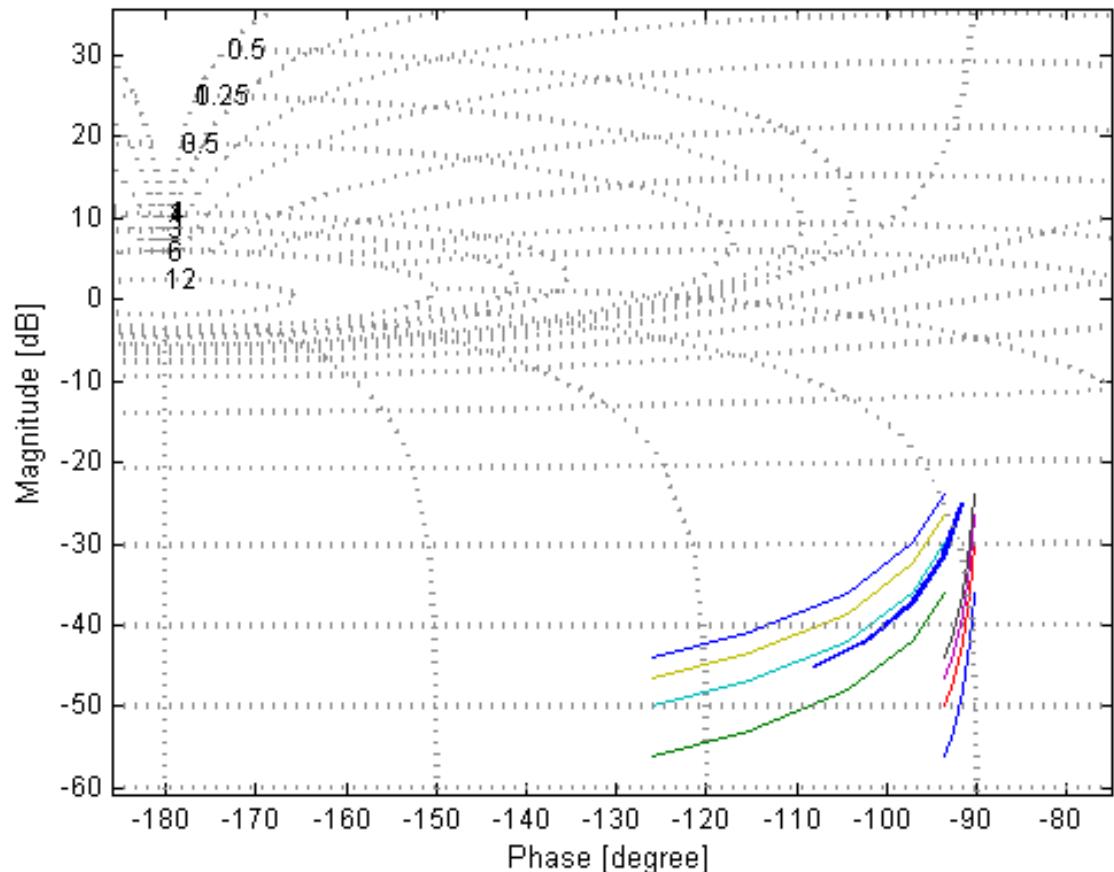
- Sensitivity specification

$$|sensitivity| \leq c(\omega)$$



Control Design Cases

- ❑ Specifications must be satisfied for all plants (for all cases)
- ❑ The identified points from the modeling on Nichols chart
- ❑ All cases can be bounded



Control Design Template

- ❑ The chosen linear plant structure:

$$\frac{k_{11}}{s} e^{-\tau_{11}s}$$

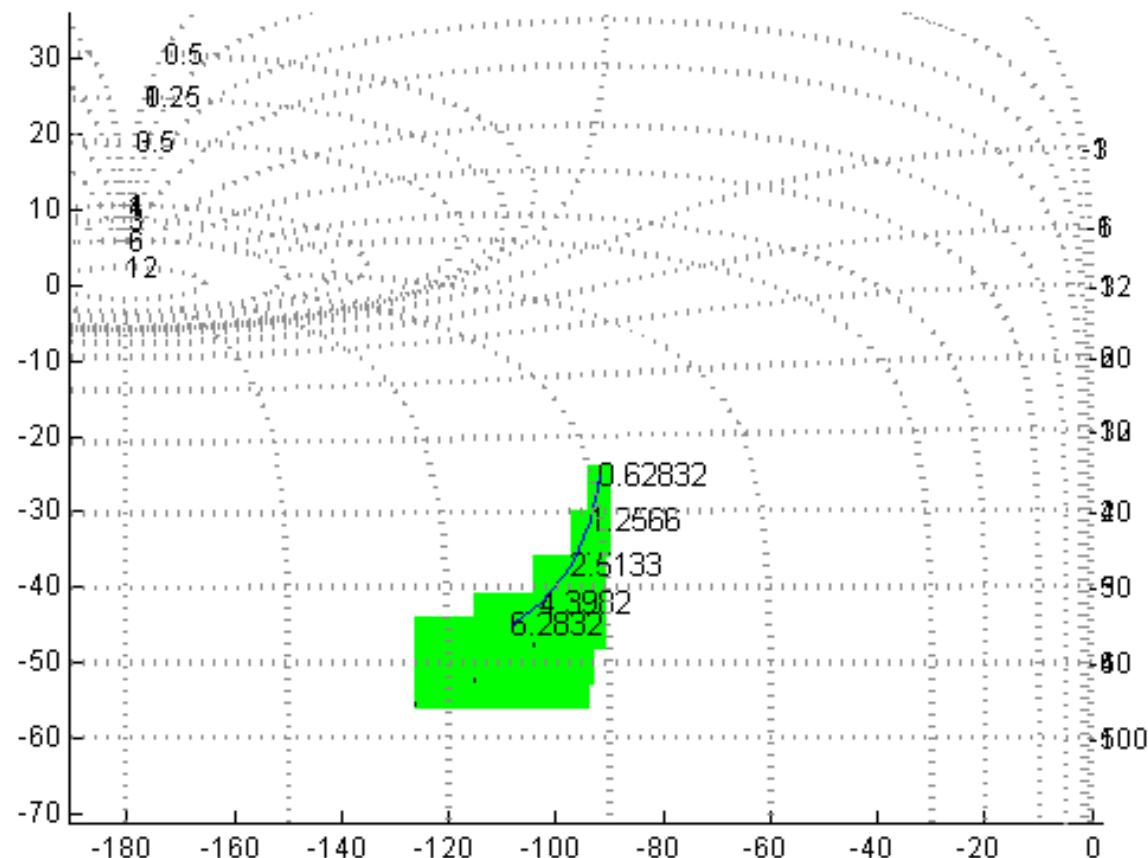
- ❑ The nominal plant is

$$\frac{0.035}{s} e^{-0.05s}$$

- ❑ The uncertainties are:

$$k_{11} \in [0.01, 0.04]$$

$$\tau_{11} \in [0.01, 0.1]$$



Control Design

Horowitz-Sidi bounds

□ Tolerance bounds

$$\left| \frac{\bar{S}_{\max}}{\bar{S}_{\min}} \right| = \frac{\max_i \left| \frac{P_i G}{1 + P_i G} \right|}{\min_i \left| \frac{P_i G}{1 + P_i G} \right|} \leq \frac{b}{a}$$

□ Sensitivity bounds

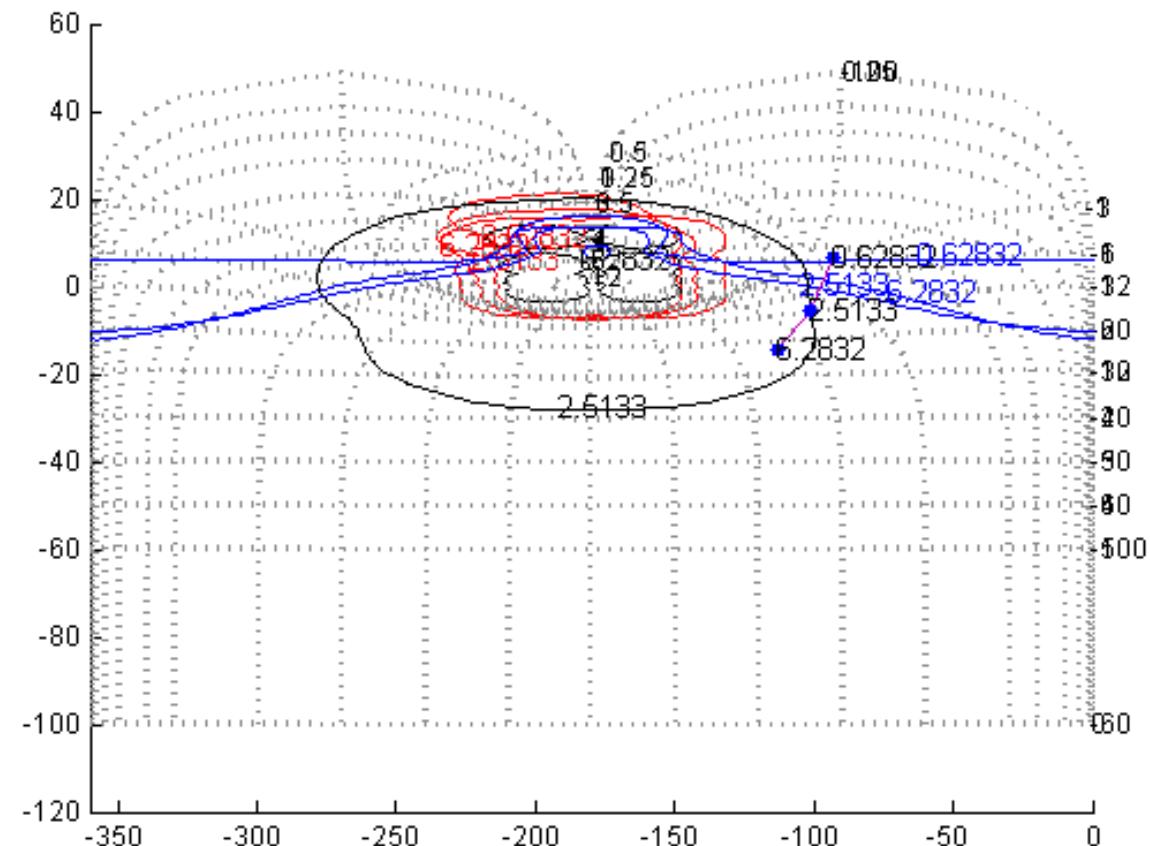
$$|S| = \left| \frac{1}{1 + PG} \right| \leq c$$

□ Cross-Coupling bounds (only for MIMO)

$$\left| \frac{\frac{W_{12}}{W_{11}} b_{22}}{1 + \frac{L_{10}}{W_{11}}} \right| \leq b_{12}$$

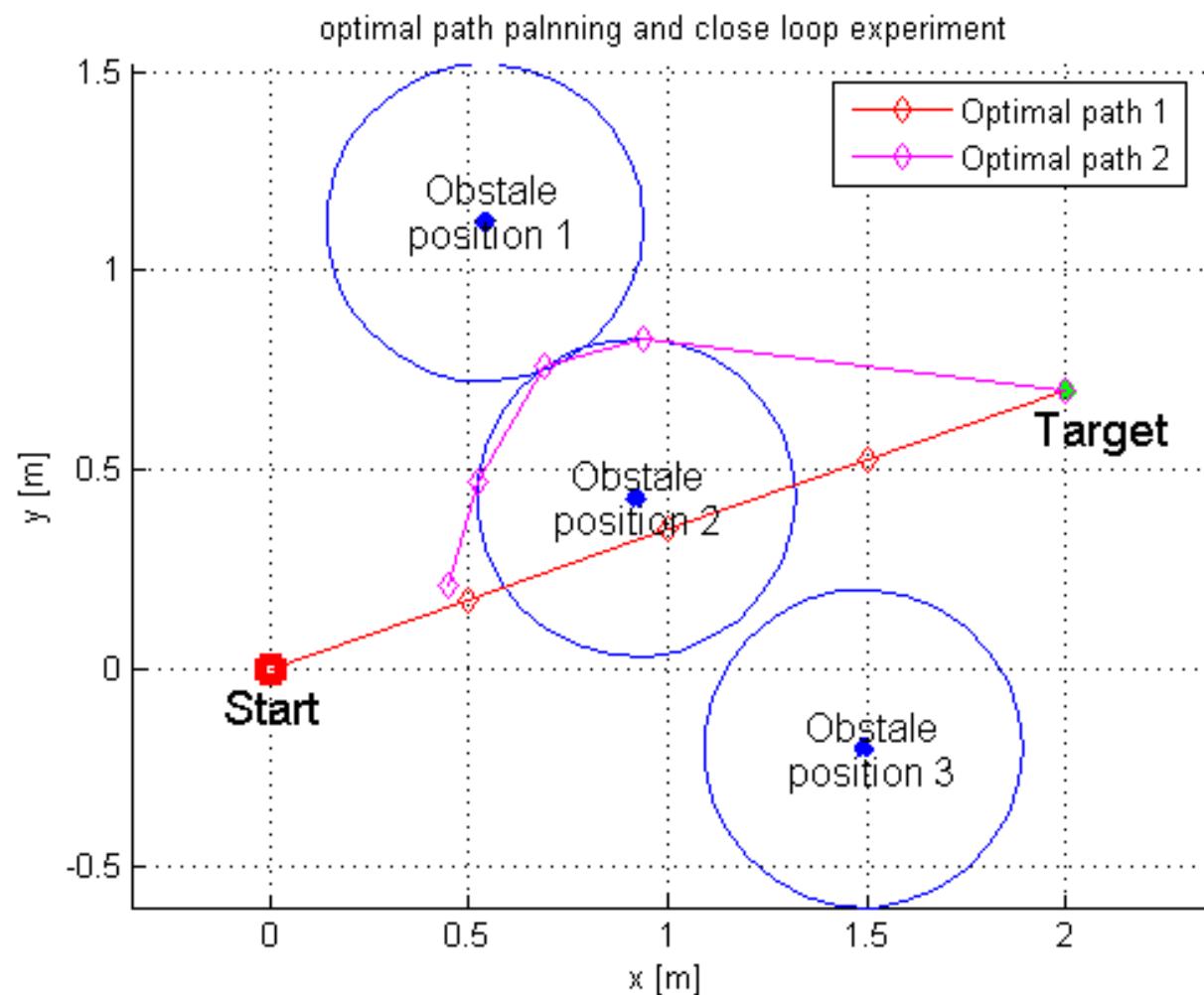
Control Design

- Loop Shaping
with the
Horowitz-Sidi
bounds
- Performance
- Sensitivity
- Cross coupling



Experiments

- ❑ Optimal Path Planner, using DIDO solver
- ❑ The obstacle is moving
- ❑ Optimal Path 2 is the updated path



Experiments

❑ Reference command tracking

❑ In this example

